

Advanced Composites
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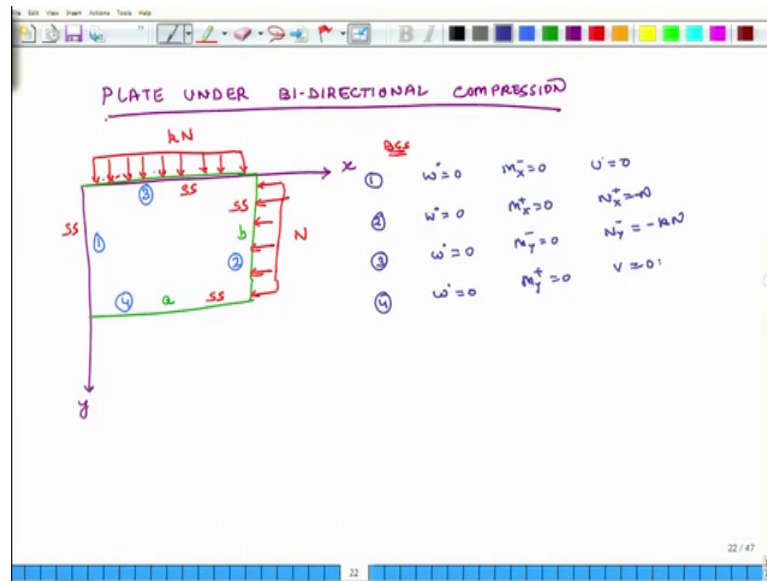
Lecture - 66
Composite Plate Under Bidirectional Compression

Hello, welcome to Advanced Composites. Today is the last day of this course and today we will continue our discussion on Buckling of Composite Plates. And in this case we will have a composite plate which will be simply supported on all 4 sides, but it will be loaded Bidirectionally. So, it will experience load and the x direction as well as in the y direction.

Now, these kinds of loads can be generated in 2 types of situations; either we are actually putting external load in both the directions. So, the external force is being applied in both the directions or these forces could also be developed. That suppose I have a plate which is being compressed in the x direction by some value n_x , and if I am constraining the v displacement in the y direction then the plate naturally will try to have an expansion as it is getting compressed in the x direction it will try to expand in the y direction due to Poisson effect.

But if I am putting constraint in the y direction so that the plate cannot expand then it will generate N_y automatically. And because of that value of N_y the plate will get compressed in bidirectionally and that will generate some different type of buckling pattern. So, in either of these cases we can solve the problem of plate buckling especially for simple geometry such as rectangular plates which are simply supported or clamped or in these kind of situations.

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So, we will look at plate under bidirectional compression. So, this is x , this is y , that is my plate; this dimension is b , this dimension is a , I mean having the external load. So, it is uniformly we getting compressed in the x direction. And here it is simply, so plate is simply supported on all 4 sides. But once again we will explicitly straight; what do the simple supports support mean.

So, it is getting compressed in this direction and also it is getting compressed in the other direction. So, let us say this is N in the x direction and the compression force resultant in the vertical direction or y direction is some constant k times N , now we look at the boundary conditions. So, when this kind of loading will happen when the v displacement is being constraints. So, if I double N , the value of N_y in the other direction we will also get doubled. So, this is a type of simulation we situation we are trying to simulate.

So, this is are so let us call these edges. So, this is edge 1; edge 2, edge 3, edge 4. So, on edge 1 we have w equals 0 M_x equals to 0, u equals to 0. On edge 2; w equals 0, M_x equals to 0, and N_x is equal to, N_x plus. On edge 3; w equals to 0, M_y equals to 0, and N_y equals k times N , and that is actually negative both are compression. And on 4th w is so these are all mid plane displacements; so w equals to 0, and y plus equals to 0, and v equals to 0.

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The whiteboard contains the following content:

- Diagram:** A rectangular plate with width a and height b . The origin $(0,0)$ is at the bottom-left corner. The x -axis is horizontal and the y -axis is vertical. The four edges are labeled with boundary conditions:
 - Left edge ($x=0$): $w=0$ (labeled 1)
 - Right edge ($x=a$): $w=0$ (labeled 2)
 - Bottom edge ($y=0$): $w=0$ (labeled 4)
 - Top edge ($y=b$): $w=0$ (labeled 3)
- Equations:**
 - At $x=0$: $w=0$, $M_x=0$, $N_x=0$
 - At $x=a$: $w=0$, $M_x=0$, $N_x=0$
 - At $y=0$: $w=0$, $M_y=0$, $N_y=0$
 - At $y=b$: $w=0$, $M_y=0$, $N_y=0$
- Text:**
 - "Understand out-of-plane behavior. : Solve only 3rd eqn."
 - "ASSUMO: $w = \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b} = w^*(x,y)$ "
 - " $w^*(x,y) = 0$ at $x=0, a$ and $y=0, b$] KINEMATICS OK"
 - "SYMMETRIC - $(B) = 0$ "
 - "ORTHOTROPIC - $D_{12} = D_{21} = 0$ "

So, once again we are interested in understanding out of plane behavior. So, we solve only the third equation, why is that the case? Because we again assume that plate is symmetric and it is also orthotropic. So, because of symmetry b is equal to 0, b matrix and because of orthotropy d_{16} is equal to d_{26} is equal to 0 ok. So, we assume a solution, so again we are going to solve this problem in an exact way.

So, we assume that the solution is $w = \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$. So, this is equal to the mid plane deflection in the z direction. So, then we see that $w = \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$ is equal to 0 at $x=0$ and $x=a$ and $y=0$ and $y=b$. So, kinematic boundary conditions are satisfied. Then we have to ensure that moment related boundary conditions on a four simply supported edges are also 0.

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$$M_x = -\left[D_{11} \frac{\partial^2 w}{\partial x^2} + D_{12} \frac{\partial^2 w}{\partial y^2}\right] = -w_0 \left[D_{11} \left(\frac{m\pi}{a}\right)^2 + D_{12} \left(\frac{n\pi}{b}\right)^2 \right] \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$$

$$M_y = -\left[D_{12} \frac{\partial^2 w}{\partial x^2} + D_{22} \frac{\partial^2 w}{\partial y^2}\right] = -w_0 \left[D_{12} \left(\frac{m\pi}{a}\right)^2 + D_{22} \left(\frac{n\pi}{b}\right)^2 \right] \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$$

$$M_x = 0 \text{ at } x = 0, a$$

$$M_y = 0 \text{ at } y = 0, b$$

All out-of-plane BC's are satisfied.

So, M_x is equal to for this plate $D_{11} \frac{\partial^2 w}{\partial x^2} + D_{12} \frac{\partial^2 w}{\partial y^2}$ and this entire thing is negative. And M_y is equal to minus $D_{12} \frac{\partial^2 w}{\partial x^2} + D_{22} \frac{\partial^2 w}{\partial y^2}$. So, this is us minus $D_{11} \left(\frac{m\pi}{a}\right)^2 + D_{12} \left(\frac{n\pi}{b}\right)^2 \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$.

And then of course, there is a w naught here and minus and this is minus w naught $D_{11} \left(\frac{m\pi}{a}\right)^2 + D_{12} \left(\frac{n\pi}{b}\right)^2 \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$. So, if I plug in the values of x and y for different edges I see that M_x is equal to 0 at x is equal to 0 and a and M_y is equal to 0 at y equals to 0 and b . So, all BC's are satisfied, out of plane BC's are satisfied.

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SATISFY PDE (3rd). $N_x = -N$ $N_y = -Nk$
 $N_{xy} = 0$

$$N_x \frac{\partial^2 w}{\partial x^2} + N_y \frac{\partial^2 w}{\partial y^2} + \frac{\partial^2 M_x}{\partial x^2} + 2 \frac{\partial^2 M_{xy}}{\partial x \partial y} + \frac{\partial^2 M_y}{\partial y^2} = 0$$

$$\rightarrow -\left[N \frac{\partial^2 w}{\partial x^2} + kN \frac{\partial^2 w}{\partial y^2} + D_{11} \frac{\partial^4 w}{\partial x^4} + 2(D_{12} + 2D_{66}) \frac{\partial^4 w}{\partial x^2 \partial y^2} + D_{22} \frac{\partial^4 w}{\partial y^4} \right] = 0$$

So, the last thing is that we satisfy the PDE, Partial Differential Equation and which partial differential equation, which is the third. So, now here we have both N_x and N_y and we know that N_x is constant in the entire plate because uniformly loaded. So, N_x does not change with respect to x and y and N_y is also constant and that is equal to k and both of them are negative.

So, and then of course, N_{xy} is 0. So, overall differential equation is $N_x \frac{\partial^2 w}{\partial x^2}$ plus $N_y \frac{\partial^2 w}{\partial y^2}$ plus $\frac{\partial^2 M_x}{\partial x^2}$ plus $2 \frac{\partial^2 M_{xy}}{\partial x \partial y}$ plus $\frac{\partial^2 M_y}{\partial y^2}$ is equal to 0. Now, we have already so this gives us minus $N \frac{\partial^2 w}{\partial x^2}$ so this is all mid plane displacement. So, $\frac{\partial^2 w}{\partial x^2}$ with respect to x , minus k times $N \frac{\partial^2 w}{\partial y^2}$ minus.

So, I am going to take minus out and everything inside the will be the positive. So, this becomes $D_{11} \frac{\partial^4 w}{\partial x^4}$ plus, $2 D_{12}$ plus $2 D_{66} \frac{\partial^4 w}{\partial x^2 \partial y^2}$ plus, $D_{22} \frac{\partial^4 w}{\partial y^4}$ equals to 0 ok.

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$$\left(-N \frac{\partial^2 w}{\partial x^2} + kN \frac{\partial^2 w}{\partial y^2} + D_{11} \frac{\partial^4 w}{\partial x^4} + 2(D_{12} + 2D_{66}) \frac{\partial^4 w}{\partial x^2 \partial y^2} + D_{22} \frac{\partial^4 w}{\partial y^4} \right) = 0$$

Substitute $w(x, y) = w_0 \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b}$.

$$\left[-N \left(\frac{m\pi}{a} \right)^2 - kN \left(\frac{n\pi}{b} \right)^2 + D_{11} \left(\frac{m\pi}{a} \right)^4 + 2(D_{12} + 2D_{66}) \left(\frac{m\pi}{a} \right)^2 \left(\frac{n\pi}{b} \right)^2 + D_{22} \left(\frac{n\pi}{b} \right)^4 \right] w_0 \sin \frac{m\pi x}{a} \sin \frac{n\pi y}{b} = 0$$

$w_0 = 0 \quad \leftarrow \text{Pre-buckling}$

or $[\quad] = 0 \quad \leftarrow \text{Buckling}$

So, this is the differential equation and now what I do is substitute; w naught x y as what, w naught $\sin m \pi$ over a times x $\sin n \pi$ y over b . So, I get minus N $m \pi$ over a whole square minus k , $n \pi$ over b whole square plus D_{11} $m \pi$ over a to the power of 4 plus, $2 D_{12}$ plus $2 D_{66}$ $m \pi$ over a $n \pi$ over b pi square entire thing squared plus $2 D_{22}$ $N \pi$ over b Whole to the power 4 into w naught $\sin m \pi$ x over a $\sin n \pi$ y over b is equal to 0.

So, again for this equation to be true which means right left side should be is 0. So, suppose I have this entire thing in green and this is w naught. So, far this equation to whole either w naught should be 0 or this entire expression in green should be 0. So, that is my pre bulking solution and this is buckling solution pre bulking (Refer Time: 14:51).

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The slide shows a handwritten equation for the buckling load N :

$$N = \frac{\pi^2}{a^2} \left[\frac{D_{11} m^4 + 2(D_{12} + 2D_{66}) m^2 n^2 R^2 + D_{22} n^4 R^4}{m^2 + n^2 R^2 k} \right]$$

Below the equation, it is noted that $R = \frac{b}{a}$ or $R = \frac{a}{b}$.

Key observations from the slide:

- As k goes up (assuming $k > 0$), $N \downarrow$
- If k is negative, N_y is tensile, $N \uparrow$
- Mode shape depends on $m, n \rightarrow R$.

So, we look at the buckling solution so, what do you get. So, if I have to solve for N , then what I get is π^2 over a^2 times $D_{11} m^4 + 2(D_{12} + 2D_{66}) m^2 n^2 R^2 + D_{22} n^4 R^4$ divided by $m^2 + n^2 R^2 k$ ok. Where; R is equal to b is equal to a over r or R equals a over b so, that the aspect ratio the plate.

So, this is my this is how I can calculate the value of N at which it is going to fail, not fail it is going to buckle. Now there are two important things one is this parameter k . So, as k goes up assuming k is more than 0 which means the compression load on the y axis also is increasing, so as k goes up N decreases and vice versa ok. Second if k is negative which means that I am pressing the plate like this, but on these side I am not pressing the plate, but I am pulling it ok.

If k is negative which means; in y direction so which means that N_y is tensile. So, what does that mean that N goes up, N goes up so plate tends to become more stable so this is something important to understand ok. And the mode shape depends on M and N and which intern is related to aspect ratio with the plate related to aspect ratio with the plate.

So, that is all I wanted to discuss for today; and this concludes are discussion on Buckling. Tomorrow we will introduce a new topic and we will switch gears is starting next week. And we will start talking about Short Fibre Composites. So, that is what I plan to do today and I look forward to seeing all of you next week as well.

Thank you.