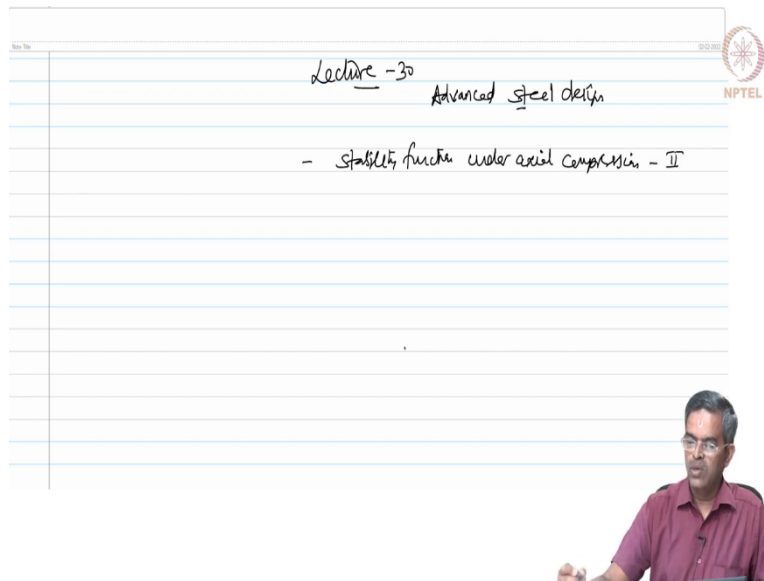


Advanced Design of Steel Structures
Dr. Srinivasan Chandrasekaran
Department of Ocean Engineering
Indian Institute of Technology, Madras

Lecture - 30
Stability functions - 2

(Refer Slide Time: 00:21)



Friends, welcome to the lecture 30 on Advanced Steel Design. We will continue to discuss Stability Functions under axial compression, we will call selection number II on the same perspective.

(Refer Slide Time: 00:52)

$$\frac{\alpha_i^2 EI}{L_i^2} \left(\frac{dy}{dx} \right) = - \left(\frac{\alpha_i}{L_i} \right) \left[k_{pp} \cot(\alpha_i) + k_{qp} \operatorname{cosec}(\alpha_i) \right] \cos\left(\frac{\alpha_i x}{L_i}\right) - \left(\frac{\alpha_i}{L_i} \right) k_{pp} \sin\left(\frac{\alpha_i x}{L_i}\right) + (k_{pp} + k_{qp}) \frac{1}{L_i}$$

$$\frac{\alpha_i^2 EI}{L_i^2} \left(\frac{dy}{dx} \right) = k_{pp} \left[1 - \alpha_i \sin\left(\frac{\alpha_i x}{L_i}\right) - \alpha_i \cot(\alpha_i) \cos\left(\frac{\alpha_i x}{L_i}\right) \right] + k_{qp} \left[1 - \alpha_i \operatorname{cosec}(\alpha_i) \cos\left(\frac{\alpha_i x}{L_i}\right) \right]$$

In the last lecture, we derived equation 11 (a) which is expressed in terms of k_{pp} and k_{qp} . So, here on the left-hand side you will see there is a term representing the slope $\left(\frac{dy}{dx}\right)$ is the slope of the v, is it not which is given as a function of k_{pp} and k_{qp} which is given by the equation 11 (a).

(Refer Slide Time: 02:27)

$$\Rightarrow \frac{\alpha_i^2 EI}{L_i} \left(\frac{dy}{dx} \right) = k_{pp} \left[1 - \alpha_i \sin \left(\frac{\alpha_i x}{L_i} \right) - \alpha_i \cot(\alpha_i) \cos \left(\frac{\alpha_i x}{L_i} \right) \right] + k_{qp} \left[1 - \alpha_i \csc(\alpha_i) \cos \left(\frac{\alpha_i x}{L_i} \right) \right] \quad \text{--- 11 (a)}$$

@ $x=0$, $\frac{dy}{dx} = \theta_p$ which is equal to Unity

@ $x=L$, $\frac{dy}{dx} = \text{Zero}$

Substitute the above condition:

$$0 = k_{pp} \left[1 - \alpha_i \sin(\alpha_i) - \alpha_i \cot(\alpha_i) \cos(\alpha_i) \right] + k_{qp} \left[1 - \alpha_i \csc(\alpha_i) \cos(\alpha_i) \right] \quad \text{--- 12 (a)}$$

The diagram shows a beam of length L under axial compression P_a . At $x=0$, there is a rotation θ_p . At $x=L$, there is a rotation θ_{rp} . The beam is fixed at $x=0$ and $x=L$.

$$\frac{\alpha_i^2 EI}{L_i} \left(\frac{dy}{dx} \right) = k_{pp} \left[1 - \alpha_i \sin \sin \left(\frac{\alpha_i x}{L_i} \right) - \alpha_i \cot \cot (\alpha_i) \cos \cos \left(\frac{\alpha_i x}{L_i} \right) \right] + k_{qp} \left[1 - \alpha_i \csc \csc (\alpha_i) \cos \cos (\alpha_i) \right]$$

We will copy this equation to the next screen and we anyway replace this with black color. So, we will do this, 11 (a).

@ $x=0$, $\frac{dy}{dx} = \theta_p$ which is equal to unity.

@ $x=L$, $\frac{dy}{dx} = \text{Zero}$.

Substitute the above condition,

$$0 = k_{pp} \left[1 - \alpha_i \sin \sin (\alpha_i) - \alpha_i \cot \cot (\alpha_i) \cos \cos (\alpha_i) \right] + k_{qp} \left[1 - \alpha_i \csc \csc (\alpha_i) \cos \cos (\alpha_i) \right]$$

Now, we know at x equal 0, $\frac{dy}{dx} = \theta_p$ because that is how we gave the rotation. This was the beam under axial compression P_a and we gave a rotation which is unity and this was θ_p we gave.

That is how we generated k_{pp} and we say k_{qp} , δ_{rp} or k_{rp} and k_{sp} , that is how we did, is it not. So, we are given θ_p which is equal to of course, unity, but we have θ_p . So, let us say θ_p .

which is equal to it whereas, at x is equal to L the slope is 0, see here. @ $x=L$, $\frac{dy}{dx} = \text{Zero}$. So, let us substitute this condition and see what happens; substituting the above conditions.

So, let us say slope is 0. So, that is going to happen at x is equal to L . So, let us write that.

$$0 = k_{pp} \left[1 - \alpha_i \sin \sin (\alpha_i) - \alpha_i \cot \cot (\alpha_i) \cos \cos (\alpha_i) \right] + k_{qp} \left[1 - \alpha_i \csc \csc (\alpha_i) \cos \cos (\alpha_i) \right]$$

, Equation 12a.

(Refer Slide Time: 05:04)

$$0 = k_{pp} - k_{pp} \alpha_i \left[\sin \sin (\alpha_i) + \cos \cos (\alpha_i) \frac{\cos \cos (\alpha_i)}{\sin \sin (\alpha_i)} \right] + k_{qp} - k_{qp} \alpha_i \cot \cot (\alpha_i)$$

This was the equation, 12a. By simplifying,

$$0 = k_{pp} \left(1 - \alpha_i \left(\frac{1}{\sin \sin (\alpha_i)} \right) \right) + k_{qp} (1 - \alpha_i \cot \cot (\alpha_i))$$

$$k_{qp} = \left[\frac{\alpha_i - \sin \sin \alpha_i}{\sin \sin (\alpha_i) - \alpha_i \cos \cos (\alpha_i)} \right] k_{pp}$$

This was equation number 13; is very simple because you know this is expressed as cos by sin. There is sin here, take a common denominator, multiply that with 0, the denominator goes away.

So, rearranging you will get k_{qp} . So, k_{qp} is given by equation 13, that is one of the coefficient we have, rotation coefficient.

(Refer Slide Time: 09:43)

$$\textcircled{a} \quad x=0, \quad \frac{dy}{dx} = 1$$

find k_{pp} !

$$k_{pp} = \frac{EI}{L_i} \left[\frac{\alpha_i \sin(\alpha_i) - \alpha_i \cos(\alpha_i)}{2(1 - \cos(\alpha_i)) - \alpha_i \sin(\alpha_i)} \right] \quad \text{--- (14)}$$

$$\textcircled{a} \quad x = 0, \quad \frac{dy}{dx} = 1$$

$$k_{pp} = \frac{EI}{L_i} \left[\frac{\alpha_i \sin(\alpha_i) - \alpha_i \cos(\alpha_i)}{2(1 - \cos(\alpha_i)) - \alpha_i \sin(\alpha_i)} \right]$$

Similarly, next condition at x is equal to 0, the slope is 1. So, substitute this condition in the original equation and you will find now k_{pp} . Call this is equation number 14. So, now, I can express the rotation functions in terms of stiffness coefficients. Let us do that.

(Refer Slide Time: 11:24)

one can express the stiffness coeffs (k_{pp}, k_{cp}) as a fn of rotation functions (r_i, c_i)

$$k_{pp} \equiv r_i = \frac{\alpha_i \sin(\alpha_i) - \alpha_i \cos(\alpha_i)}{2(1 - \cos(\alpha_i)) - \alpha_i \sin(\alpha_i)} \quad \text{15(a)}$$

$$k_{cp} \equiv c_i = \frac{\alpha_i - \sin(\alpha_i)}{\sin(\alpha_i) - \alpha_i \cos(\alpha_i)} \quad \text{15(b)}$$

$k_{pp} = (a) \frac{EI}{L}$
 $k_{cp} = (b) \frac{EI}{L}$

(r_i, c_i) rotation function for compressed axial load case

$$k_{pp} = r_i = \left[\frac{\alpha_i \sin \sin(\alpha_i) - \alpha_i \cos \cos(\alpha_i)}{2(1 - \cos \cos(\alpha_i)) - \alpha_i \sin \sin(\alpha_i)} \right]$$

$$k_{qp} = c_i = \left[\frac{\alpha_i - \sin \sin(\alpha_i)}{\sin \sin(\alpha_i) - \alpha_i \cos \cos(\alpha_i)} \right]$$

One can express the stiffness coefficients, that is k_{pp} and k_{qp} . These are rotation coefficients; as a function of rotation functions r_i and c_i .

Friends, if you remember k_{pp} this was some function of EI by L and k_{qp} is also some function of EI by L. Here also you look at this equation k_{pp} is a function of EI by L and k_{qp} is a function of k_{pp} which is again EI by L. So, I am just trying to express this, only these k_{pp} as r_i and k_{qp} as c_i . We will call this equation as 15 (a) and 15 (b). These are called stability functions, that is rotation functions for stability.

They are called as r_i and c_i are called rotation functions for compressive axial load case. I can take a special case of axial load zero and see what happens to these functions. Let us take a special case of axial load zero and see what happens.

(Refer Slide Time: 14:58)

Special case of axial load zero ($P_a \rightarrow 0$)

$\Rightarrow \alpha_i = \text{zero}$

one need to apply L'Hospital's rule

NPTEL

Special case of axial load zero, that is P_a is set to 0. So, what does it mean? Immediately, it implies a fact that α_i will become zero. So, when α_i becomes zero, look at the rotation

(Refer Slide Time: 18:32)


$$\lim_{x \rightarrow a} \frac{f(x)}{g(x)} = \lim_{x \rightarrow a} \left(\frac{f'(x)}{g'(x)} \right) \text{ if R.H.S exists}$$

for ex

$$\lim_{x \rightarrow 0} \frac{e^x - 1}{x^2 + x} = \lim_{x \rightarrow 0} \frac{\frac{d}{dx}(e^x - 1)}{\frac{d}{dx}(x^2 + x)} = \lim_{x \rightarrow 0} \frac{e^x}{2x + 1} = 1$$

L'Hospital's Rule uses derivatives to evaluate the limits involving indeterminate forms

- for $\frac{0}{0}$ or $\frac{\infty}{\infty}$ limit of form is equal to limit of derivative.



Limit x tends to 0, $f(x)$ by $g(x)$ is equal to limit x tends to a f dash of x g dash of x , if the right hand side of the equation exists. Let us say for example, limit x tends to 0 e power x minus 1 by x square plus x is our 0 by 0 form, but the denominator derivative is it not 0, is it not it exists. So, I can say is equal to limit x tends to 0 d by dx of the numerator d by dx of the denominator which is going to be limit x tends to 0 e power x by $2x + 1$. Now, the answer will be 1, for which the condition is the right-hand side should exist.

So, L' Hospital rule uses derivatives to evaluate the limits involving indeterminate forms. It states that for indeterminate functions, where the unity tends to form 0 by 0 or infinity by infinity, the limit of that form is equal to the limited derivative itself.

For 0 by 0 form or this form, limit of the form is equal to limit of the derivative. L' Hospital rule can be applied any number of times until the function does not reduce to a condition back again to 0 by 0 or infinity by infinity. So, we have a problem which is to be used in this specific case. So, let us consider the rotation functions 15 (a) and 15 (b).

(Refer Slide Time: 22:07)


Let us consider Eq 15(a) & 15(b)

$$k_{pp} \equiv r_i = \left[\frac{\alpha_i \sin(\alpha_i) - \alpha_i \cos(\alpha_i)}{2(1 - \cos(\alpha_i)) - \alpha_i \sin(\alpha_i)} \right] \quad 15(a)$$

$$k_{qp} \equiv c_i = \left[\frac{\alpha_i - \sin(\alpha_i)}{\sin(\alpha_i) - \alpha_i \cos(\alpha_i)} \right] \quad 15(b)$$

As $\alpha_i \rightarrow 0$, (no axial load case)
 both $\frac{f(\alpha_i)}{g(\alpha_i)}$ approaches zero - (0/0) form

$$r_i @ \alpha_i = 0 = 4$$

$$c_i @ \alpha_i = 0 = 0.5$$


$$k_{pp} = r_i = \left[\frac{\alpha_i \sin(\alpha_i) - \alpha_i \cos(\alpha_i)}{2(1 - \cos(\alpha_i)) - \alpha_i \sin(\alpha_i)} \right]$$

$$k_{qp} = c_i = \left[\frac{\alpha_i - \sin(\alpha_i)}{\sin(\alpha_i) - \alpha_i \cos(\alpha_i)} \right]$$

So, let us copy those functions here. Let us rub this. As α_i reduces to zero or approaches zero because we are considering no axial load case, is not that; both $f(\alpha_i)$ by $g(\alpha_i)$ approaches zero. So, this will turn to a 0 by 0 form, is it not. So, I can apply L' Hospital rule which will give me the function value as r_i will now become that is r_i at α_i equal 0 will become 4 and c_i at α_i becomes 0 is 0.5.

(Refer Slide Time: 24:13)

@ $P_a = 0$, the stiffness coeffs will reduce to the conventional c/o factors

Now, after substituting


$$k_{pp} = r_i \frac{EI}{L_i}$$

$$k_{qp} = c_i k_{pp} = c_i r_i \left(\frac{EI}{L_i} \right)$$

$$k_{rp} = \frac{k_{pp} + k_{qp}}{L_i} = \frac{r_i EI}{L_i^2} (1 + c_i)$$

$$k_{sp} = -k_{rp}$$

16



So, at P_a equals 0, the stiffness coefficients will reduce to the conventional carryover factors.

$$k_{pp} = r_i \frac{EI}{L_i}$$

$$k_{qp} = c_i k_{pp} = c_i r_i \left(\frac{EI}{L_i} \right)$$

$$k_{rp} = \frac{k_{pp} + k_{qp}}{L_i} = \frac{r_i EI}{L_i^2} (1 + c_i)$$

$$k_{sp} = -k_{rp}$$

Let us call this equation as equation 16. Similarly, friends by applying any rotation the kth end, another set of stiffness coefficients can be derived.

(Refer Slide Time: 25:58)

By applying unit rotation @ k^{th} end, another set of stiffness coeffs can be derived.

$k_{pq} = c_i r_i \frac{EI}{L_i}$
 $k_{qq} = r_i \frac{EI}{L_i}$
 $k_{rq} = ?$
 $k_{sq} = ?$

$$k_{pq} = c_i r_i \frac{EI}{L_i}$$

$$k_{qq} = r_i \frac{EI}{L_i}$$

Now, by applying unit rotation of the k^{th} end, another set of stiffness coefficients can be derived. Let us draw the figure and understand that I am trying to apply unit rotation at the k^{th} end. So, this is θ_q which is unity and we know the length of the beam is L_i and we have already applied the axial load P_a . And, this is k_{pq} , this is k_{qq} and this force of coefficient is k_{rq} and this is k_{sq} .

So, I can again cut a section here, draw the free body diagram. P_a is applied here and P_a is reciprocated here. And, there is a moment k_{qq} and there is force k_{sq} which is actually equal to $-\frac{k_{pq} + k_{rq}}{L_i} k_{qq}$. Why it is minus? I think you realize it there is a net moment of $k_{pq} + k_{rq}$. Here, the counteract that this will be the couple and this is opposite to k_{sq} . Therefore, there is a minus sign here.

So, you can follow the same logic and we can write the coefficients directly as here. I am leaving it for learning, you easily do that.

(Refer Slide Time: 29:49)

Summary

Stability fn. (rotational)
(c_i, r_i)

for

$P_a = f(P_e)$

i) axial comp load (P_a)
ii) when $P_a \rightarrow 0$.

So, friends we have learnt the derivation of stability function, that is rotation coefficients which are c_i and r_i for two cases. Case i, under axial compressive load P_a , ii when P_a tend to 0 and we realize that in both the cases P_a is expressed as a function of PE. Why? We are looking for stability functions. So, please look at the derivation back again and try to understand that is very simple.

It has got involvement of differential equation understanding, have a parallel reading on some text book on engineering mathematics. And, learn this and try to get a hold of the derivation. The next lecture we are going to discuss about the rotation function and stability functions of a beam under axial tensile load.

Thank you very much friends. Have a good day.