

Advanced Design of Steel Structures
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Lecture - 36
Unsymmetric bending - 1

(Refer Slide Time: 00:18)

Lecture 36
Adv steel design
- Unsymmetric bending - 1

In symmetric bending, bending takes place in a plane parallel to the plane of applied moment

M_z Moment about z axis
- whose trace is seen in the xy plane
In the x-section of the beam, y-y axis is the trace of applied moment

Friends, welcome to the 36th lecture on Advanced Steel Design, where we are going to learn something about Unsymmetric Bending. So, we call unsymmetric bending, lecture 1. In the previous lectures, we have learnt the importance of plastic design form resistant design, material properties at elevated temperature for steel, newly proposed materials as functionally graded materials and their advantages.

We also did some examples on plastic design and analysis, then for form dominant design we understood that stability which is one of the modes of failure is important to assess and we developed stability functions for various application conditions of axial compressive load, axial tensile load and etcetera and we did some problems and used MATLAB intensively to find out the critical buckling load so that I can assess stability of this systems.

Now, unsymmetric bending is another area of challenge in advanced design procedures. Generally, friends, in symmetric bending, bending takes place in the plane parallel to the plane of applied movement. Let us try to explain this with some graphical illustration. So, let

us take a beam. So, let us say this is my x-axis of the beam which is along the span of the beam and let us call this as y and you know the z-axis is normal to this laptop screen.

So, with that axis I apply a moment about that axis I call that as M_z . So, M_z is the moment about z-axis whose trace is seen on the XY plane, right. So, now, if I cut a section, if I cut a section and if the section cross-section is known to me where this becomes my x-axis now and I call this as y and this as z-axis, correct? Imagine this figure 1 and figure 2. Figure 1 positive z is normal to the laptop screen, look it from the side.

So, you will see y in the vertical direction and z towards left side, am I right. So, I think I will remove this write it better here and remove this maybe write it here. So, now, I can say a about in the cross-section of the beam y-y axis is the trace of applied moment; z-z axis is the axis about which moment is applied. We call that is y as M_z .

(Refer Slide Time: 05:21)

The slide contains handwritten notes on lined paper. At the top right is the NPTEL logo. The notes are as follows:

- Bending takes place in a plane \perp to the plane of applied moment
- Moment about y-y axis is kept as ZERO
- Mathematically, $\sum M_y = 0$. (1)

Hence, there are no external moments acting about y-y axis
the internal resisting moment about y-y axis should be ZERO

$$\sum M_y = \int_A \sigma_x dA \cdot y \equiv 0 \quad (2)$$

We also know, $\frac{\sigma}{y} = \frac{E}{R}$ Hence $\frac{E}{R} \int_A \sigma_x dA = 0$. (3)

In the bottom right corner, there is a small video inset of a man in a light blue shirt speaking.

Having said this, we can now write a condition saying bending takes place in a plane parallel to the plane of applied moment.

So, therefore, friends we know that moment about y-y axis is kept as zero. Mathematically algebraic sum of moments about y is 0, we call equation number 1. Therefore, there are no external moments acting about y-y axis, right. Since there are no external moments acting about y-axis the internal resisting moment about y-y axis should be 0.

So, can I say mathematically M_y should be integral of over area A $\sigma \times dA$, 0 equation 2.

So, now, we also know

$$\frac{\sigma}{y} = \frac{E}{R}$$

, hence $\frac{E}{R}$ into $\sigma \times dA$ of the area a should be equal to 0. So, now, $\sigma \times$ is the product of moment of inertia.

(Refer Slide Time: 08:18)

The slide contains handwritten notes on a lined background. At the top right is the NPTEL logo. The main text is as follows:

$$\equiv \int_A (zy) dA = 0 \quad \text{---} \quad (3a)$$

The above Eqn (3a) is true only when $(z-z)$ & $(y-y)$ axes are principal axes.

Hence, for a symmetric bending,

- 1) It is essential that the plane contains
 - a) principal axes of Inertia
 - b) plane of applied moment
 - c) plane of deflection

| should coincide

In the bottom right corner, there is a small video inset showing a man in a light blue shirt and glasses, likely the lecturer.

So, let us do that way saying it should be same as over the area A I should say $z y dA$ should be termed as 0, we call this equation 3. Do not mistake it as stress it is not stress to ensure this condition, we should imply that they should be satisfied. So, we can say $z y dA$ should be termed as 0. We call this 3(a) the above equation is only true when $Z - Z$ and $Y - Y$ axis are principal axis.



The above equation 3(a) is true only when $Z - Z$ and $Y - Y$ axes or principal axes. Hence for the symmetric bending there are certain conditions. 1 - it is essential that the plane containing a - principal axes of inertia; b - plane of applied moment; c - plane of deflection should coincide, then only a symmetric bending can happen.

(Refer Slide Time: 10:51)

(2) It is also obvious that Neutral Axis will coincide with other principal axes of Inertia

When, the trace of plane of applied moment does not coincide with any of the principal axes of Inertia, then it is called Unsymmetric bending.

This is also called as non-uniplanar bending.





The second condition for the symmetric bending to happen is, it is also obvious that the neutral axis will coincide with other principal axes of inertia. So, therefore, friends, when the trace of plane of applied moment does not coincide with any of the principal axes of inertia, then it is called unsymmetric bending.

This is also called by another name which is non uniplanar bending. This is another name of unsymmetric bending.

(Refer Slide Time: 12:55)

What are the consequences if a member is subjected to unsymmetric bending?

- 1) NA will not be \perp to the trace of the applied moment-plane
- 2) Deflection curve will not be a simple plane curve in the plane of applied moment
- 3) Members, which are not symmetric about the vertical axis & have a special x-axes, then these members will undergo TWISTING under transverse loads
- 4) To avoid twisting of unsymmetric section, load should be applied @ the shear centre



Now, the question comes if a member or a cross-section is subject to unsymmetric bending what will be the consequence?

So, we will ask a question what are the consequences if a member is subjected to unsymmetric bending. 1st, the neutral axis will not be perpendicular to the trace of the applied moment plane it makes an inclination. 2 – The deflection curve is also not a plane curve or we should say will not be a simple plain curve in the plane of applied moment.

3 – members which are not symmetric because you may get members which are not symmetric in cross-section. So, members which are not symmetric about the vertical axis and have a special cross-section, then these members will undergo twisting under transverse loads, that is a very serious consequence we have when the members are subjected to unsymmetric bending.

The 4th could be to avoid twisting of unsymmetric section load should be applied at the shear center.

(Refer Slide Time: 16:33)

Q) what is the speciality in analysis if a member is subjected to unsymmetric bending?

Applied Moment, in such cases will cause bending about both the principal axes of Inertia

Hence, stress @ any section, or @ any point in the X-Y plane cannot be determined using the classical theory of bending

$$\frac{E}{R} = \frac{\sigma}{y} = \frac{M}{I} \quad \times \text{ can't be used for unsym. b.}$$

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Now, the question comes what is a specialty in analysis if a member is subjected to unsymmetric bending. The answer is the applied moment, in such cases will cause bending about both the principal axes of inertia. Therefore, stress at any section or at any point in the cross section cannot be determined using the classical theory of bending that

$$\frac{E}{R} = \frac{\sigma}{y} = \frac{M}{I}$$


This equation cannot be used to determine the stress. So, it is very complicated problem now.

(Refer Slide Time: 18:41)

4) How to determine the stress @ any point in the x-sections of a member under unsymmetric bending?

It is necessary to transform the problem of unsymmetric bending into Uni-planar bending.

5) How to do this transformation?



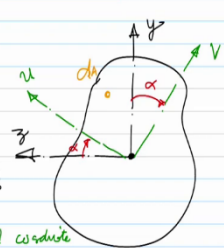
Now, what do I do to determine the stress when a member is subjected to unsymmetric bending?

So, the question is then how to determine the stress at any point in the cross-section of a member under unsymmetric bending. To do that, it is necessary to transform the problem of unsymmetric bending into uni-planar bending. So, we need to do a transformation of the problem.

Now, the question comes how will you transform the problem into uni-planar bending? that is a interesting question because we need to find the stresses at the cross sections.

(Refer Slide Time: 20:16)

To transform Unsymmetric bending to Uniplanar bending



U-V axis is inclined (α) to z-axis
 α - measured from z-axis, clockwise to mark U-V axis
 α is measured in the +ve V coordinate

from Fig(1), we can write the follg Eqn

$$\begin{aligned} U &= z \cos \alpha + y \sin \alpha \\ V &= -z \sin \alpha + y \cos \alpha \end{aligned} \quad (1)$$

Fig(1) x-y-z

Let us look into that. So, the objective is now to transform unsymmetric bending problem to uni-planar bending that is objective now. Let us take a cross-section. Let us mark these two axes as z and y; x is normal to the screen. Let us also mark two more additional axis which is called the v and u axis.

$$u = z \cos \alpha + y \sin \alpha$$

$$v = -z \sin \alpha + y \cos \alpha$$

Let us measure the angle of u-axis from z-axis or v-axis from y-axis as α . Let us take a point whose area is dA , it has got specific coordinates u and v or x, y depending upon our own choice of coordinates. Let us refer this as figure one a typical cross section of a member.

One can very well see here that u axis is inclined α to z axis α is measured from z - z axis clockwise to mark u - u axis, that is a condition here. So, to be very specific α is measured in the positive V coordinate. So, we have positive V , we measure from that. So, from this figure I can write a basic relationship which is very easy write down that. So, from figure 1, we can write the following equation.

What are they? $u = z \cos \alpha + y \sin \alpha$. We call this equation number 1; it is a basic equation written with reference to figure number. having said this, I want to find the moment of inertia.

(Refer Slide Time: 24:27)


To find MoI (uu & vv) axes

from first principles, we know

$$I_u = \int_A v^2 dA$$

$$I_v = \int_A u^2 dA$$

2a

$$I_u = \int_A [(-z \sin \alpha) + (y \cos \alpha)]^2 dA$$


Let us say to find the moment of inertia about uu and vv axes remember they are inclined to zz and yy. So, it is not that easy to use a principle parallel axes or perpendicular axes theorem to find out that. They are inclined by an angle α . So, it is not that easy to find out I_{uu} and I_{vv} .

So, let us use first principles. So, from first principles, we know I_u is integral $v^2 dA$ for the whole area A. I_v is integral $u^2 dA$ for the whole area a. We call this equation as 2(a).

Now, we have expression for u and v? Do we have expressions for u and v, let us substitute it here and see what happens. So, now, I_v becomes integral over area A u is z of $\cos \alpha$, v is minus z $\sin \alpha$. So, minus z $\sin \alpha$ plus y $\cos \alpha$ the whole square dA. So, I am getting I_u , I am getting I_u . So, v is here I am using this relationship. Let us expand this.

(Refer Slide Time: 26:36)

The slide shows a handwritten derivation for the moment of inertia I_u . At the top left is a menu bar with options like Copy, Paste, Select All, etc. At the top right is the NPTEL logo. The main content is as follows:

$$I_u = \int_A (z^2 \sin^2 \alpha + y^2 \cos^2 \alpha - 2zy \sin \alpha \cos \alpha) dA$$

$$= \sin^2 \alpha \int_A z^2 dA + \cos^2 \alpha \int_A y^2 dA - \sin 2\alpha \int_A zy dA$$

($\sin 2\alpha = 2 \sin \alpha \cos \alpha$)

We also know from the first principle that

$$\int_A z^2 dA = I_y \quad \Bigg| \quad I_u = I_y \sin^2 \alpha + I_z \cos^2 \alpha - I_{yz} \sin 2\alpha$$

$$\int_A y^2 dA = I_z \quad \Bigg| \quad \text{--- (2)}$$

At the bottom right of the slide, there is a small video inset showing a man in a light blue shirt sitting at a desk.

So, let us say I_u is equal to integral z square \sin square α plus y square \cos square α minus $2zy \sin \alpha \cos \alpha$ of dA . So, which will be \sin square α of z square dA whole area A plus \cos square α of y square dA whole area A minus $\sin 2\alpha$ integral zy dA .

How do we write this? We know $\sin 2\alpha$ is $2 \sin \alpha \cos \alpha$. We also know from the first principles that z square dA for the whole area A can be said as I_y integral y square dA for the whole area A can be said as I_z when we know this then I can write I_u from the above equation as,

$$I_u = I_y \sin^2 \alpha + I_z \cos^2 \alpha - I_{yz} \sin 2\alpha$$

We will call this equation number 2.

Let us simplify this. Let us copy this equation here. I will use a tool. I will copy this equation.

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Handwritten derivation on a slide:

$$I_u = I_y \sin^2 \alpha + I_z \cos^2 \alpha - I_{yz} \sin 2\alpha \quad (2)$$

We know,

$$\cos 2\alpha = 1 - 2\sin^2 \alpha$$

$$= 2\cos^2 \alpha - 1$$

$$\therefore \sin^2 \alpha = \frac{1 - \cos 2\alpha}{2}$$

$$\cos^2 \alpha = \frac{1 + \cos 2\alpha}{2}$$

$$I_u = \frac{I_y}{2} (1 - \cos 2\alpha) + \frac{I_z}{2} (1 + \cos 2\alpha) - I_{yz} \sin 2\alpha$$

$$I_u = \frac{I_y + I_z}{2} + \frac{I_z - I_y}{2} \cos 2\alpha - I_{yz} \sin 2\alpha$$

$$I_u = \frac{I_y + I_z}{2} + \frac{I_z - I_y}{2} \cos 2\alpha - I_{yz} \sin 2\alpha \quad (3)$$

Let me copy it here. So, I am going to simplify this equation. Let me remove this number.

$$I_u = I_y \sin^2 \alpha + I_z \cos^2 \alpha - I_{yz} \sin 2\alpha$$

So, I am going to replace these two values with these new expressions.

$$\cos \cos 2\alpha = 1 - 2\sin^2 \alpha$$

$$\cos \cos 2\alpha = 2\cos^2 \alpha - 1$$

$$\sin^2 \alpha = \frac{1 - \cos \cos 2\alpha}{2}$$

$$\cos^2 \alpha = \frac{1 + \cos \cos 2\alpha}{2}$$

$$I_u = \frac{I_y}{2} (1 - \cos \cos 2\alpha) + \frac{I_z}{2} (1 + \cos \cos 2\alpha) - I_{yz} \sin 2\alpha$$

$$I_u = \frac{I_y + I_z}{2} + \frac{I_z - I_y}{2} \cos \cos 2\alpha - I_{yz} \sin 2\alpha$$

call this equation number 3. So, let us take equation number 3 as one of the important reference derivatives.

(Refer Slide Time: 32:19)

To find I_v

$$I_v = \int_A u^2 dA$$

$$= \int_A (z \cos \alpha + y \sin \alpha)^2 dA$$

$$= \int_A (z^2 \cos^2 \alpha + y^2 \sin^2 \alpha + 2zy \sin \alpha \cos \alpha) dA$$

$$I_v = \cos^2 \alpha \int_A z^2 dA + \sin^2 \alpha \int_A y^2 dA + \sin 2\alpha \int_A zy dA$$

$$I_v = I_y \cos^2 \alpha + I_z \sin^2 \alpha + I_{yz} \sin 2\alpha$$

Now, let us move on to I_v . Now, let us find I_v . We know I_v is integral $u^2 dA$ for the entire A ; we already know u as this value, $u = z \cos \alpha + y \sin \alpha$. Let us substitute it here which is going to be equal to integral for the whole area $z \cos \alpha + y \sin \alpha$ the whole square dA . Let us expand this which will be $z^2 \cos^2 \alpha + y^2 \sin^2 \alpha + 2zy \sin \alpha \cos \alpha$ of dA integrated whole area A .

Now, I can write I_v as $\cos^2 \alpha$ integral $z^2 dA$ for the whole area A plus $\sin^2 \alpha$ integral $y^2 dA$ for the whole area plus $\sin 2\alpha$ integral whole area $zy dA$. I think there is no confusion in writing this expression. Now, I can write this as $z^2 dA$ is $I_y \cos^2 \alpha$, $y^2 dA$ is $I_z \sin^2 \alpha$ plus this is going to be $I_{yz} \sin 2\alpha$. This is I_v . Now, let us take this I_v and expand it.

$$I_v = I_y \cos^2 \alpha + I_z \sin^2 \alpha + I_{yz} \sin 2\alpha$$

We have a substitution.

(Refer Slide Time: 34:41)

$$I_v = (1 + \cos 2\alpha) \frac{I_y}{2} + (1 - \cos 2\alpha) \frac{I_z}{2} + I_{yz} \sin 2\alpha$$

$$I_v = \frac{I_y + I_z}{2} - \frac{I_z - I_y}{2} \cos 2\alpha + I_{yz} \sin 2\alpha \quad (4)$$

$$I_u = \frac{I_y + I_z}{2} + \frac{I_z - I_y}{2} \cos 2\alpha - I_{yz} \sin 2\alpha \quad (3)$$
 add (3) & (4),

$$I_u + I_v = I_y + I_z \quad (5)$$



$$I_v = (1 + \cos 2\alpha) \frac{I_y}{2} + (1 - \cos 2\alpha) \frac{I_z}{2} + I_{yz} \sin 2\alpha$$

$$I_v = \frac{I_y + I_z}{2} - \frac{I_z - I_y}{2} \cos 2\alpha + I_{yz} \sin 2\alpha$$

$$I_u = \frac{I_y + I_z}{2} + \frac{I_z - I_y}{2} \cos 2\alpha - I_{yz} \sin 2\alpha$$

$$I_u + I_v = I_y + I_z$$



I get this classical relationship, I call this equation number 5. It is a classical relationship.

(Refer Slide Time: 37:31)

To find I_{uv}

$$I_{uv} = \int_A uv \, dA$$

Sub for (u, v) from Eq. 1.

$$= \int_A (z \cos \alpha + y \sin \alpha) (-z \sin \alpha + y \cos \alpha) \, dA$$
$$= \int_A (-z^2 \sin \alpha \cos \alpha - yz \sin^2 \alpha + yz \cos^2 \alpha + y^2 \sin \alpha \cos \alpha) \, dA$$


Let us also find out the cross product which is I_{uv} . We know I_{uv} is given by integral $u \cdot v \, dA$ over the area A . So, we already have the values for u and v ; let us see here in equation. We have equation 1, we have u and v . So, let me write it here. So, substitute for u and v from equation 1.

So, let us do that which will be equal to integral $z \cos \alpha + y \sin \alpha$ into $-z \sin \alpha + y \cos \alpha$ of dA for the whole area which now become integral $-z^2 \sin \alpha \cos \alpha - yz \sin^2 \alpha + yz \cos^2 \alpha + y^2 \sin \alpha \cos \alpha$. This is for the whole area.

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$$I_{uv} = \int_A (z^2 \sin \alpha \cos \alpha - yz \sin^2 \alpha + yz \cos^2 \alpha + y^2 \sin \alpha \cos \alpha) dA$$

$$I_{uv} = -\sin \alpha \cos \alpha \int_A z^2 dA + \sin \alpha \cos \alpha \int_A y^2 dA + \int_A (yz \cos^2 \alpha - yz \sin^2 \alpha) dA$$

$$= -I_y \sin \alpha \cos \alpha + I_z \sin \alpha \cos \alpha + I_{yz} (\cos^2 \alpha - \sin^2 \alpha)$$

$$I_{uv} = \frac{(I_z - I_y)}{2} \sin 2\alpha + I_{yz} \cos 2\alpha \quad (6)$$

Let me copy this equation take it to the next screen for our reference which is now this is I_{uv} . So, I_{uv} is now. So, we have two terms $\sin \alpha \cos \alpha$ twice. So, what I will do that is I can write $\sin \alpha \cos \alpha$ integral z square dA , let us have a minus sign plus $\sin \alpha \cos \alpha$ integral y square dA plus yz dA . we will do it. This way plus yz \cos square α minus yz \sin square α dA for the whole area A .


So, now I can expand this further. So, we know this is minus of I_y $\sin \alpha \cos \alpha$ and this is I_z . So, plus I_z of $\sin \alpha \cos \alpha$ plus you know yz is A dA is I_{yz} , \cos square α minus \sin square α , am I right? Which now amounts to I_z minus I_y of $\sin 2\alpha$ by 2 plus I_{yz} $\cos 2\alpha$, equation 6. This is my I_{uv} it is a classical equation.

$$I_{uv} = \frac{(I_z - I_y)}{2} \sin 2\alpha + I_{yz} \cos 2\alpha$$

Friends, we must subsequently substitute I_{uv} to be 0. So, now subsequently we should substitute I_{uv} as 0. Why do we have to do that? In simple terms, I_{uv} need to be indicated as an understanding of explaining non-symmetric bending.

(Refer Slide Time: 42:50)

for (u-v) & (w-v) axes being principal axes of Inertia,

$$I_{uv} = 0.$$
$$I_{uv} = \frac{(I_z - I_y)}{2} \sin 2\alpha + I_{yz} \cos 2\alpha \quad \text{--- (6)}$$
$$= 0; \quad \tan (2\alpha) = - \left(\frac{2I_{yz}}{I_z - I_y} \right) \quad \text{--- (7)}$$


$$I_{uv} = 0;$$

$$I_{uv} = \frac{(I_z - I_y)}{2} \sin 2\alpha + I_{yz} \cos 2\alpha$$

$$\tan (2\alpha) = - \left(\frac{2I_{yz}}{I_z - I_y} \right)$$

So, now friends, u and v being principal axes of inertia I_{uv} should be 0. So, what is I_{uv} ? I am just copying this equation. So, if you equate this to 0, I will get $\tan (2\alpha) = - \left(\frac{2I_{yz}}{I_z - I_y} \right)$. We call this equation number 7. So, friends we are now interested to find out stress at any point in a given cross section.

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The slide contains handwritten text on a lined background. At the top, it asks 'To find (σ_P) at any point in the x-section?'. Below this is a section titled 'Summary' with three bullet points: 'Unsymmetric bending', 'Consequences?', and ' (σ_P) , how do we calculate'. To the right of these points are the terms I_u , I_v , $I_{uv} = 0$, and 'To find α '. In the bottom right corner, there is a small video inset of a man in a light blue shirt speaking.

So, our objective is to find stress at any point in the cross-section. That is objective. So, we will look into this in detail in the next lecture.

Let us now write the summary what we learnt in this lecture. We learnt what is meant by unsymmetric bending or non-uniplanar bending. We have also learnt what are the consequences if a member is subjected to unsymmetric bending. If I want to find the stress at any point P in a cross section or a member under unsymmetric bending, what do I do what do we or how do we calculate. So, we said that we need to convert the unsymmetric bending problem into a uniplanar bending problem.

To do that we need to find the principal moments of inertia and we equated the product to 0 being principal axis and to find the inclination of α . Remember α is an angle is an angle of u-u axis with z-z axis measuring from the positive z to positive u or positive v to positive y, in a specific manner. So, if I know this using the equation, we will be able to find the stresses. That is what we discussed briefly and we will explain the equation how to find stresses in case of an unsymmetric bending in the next lecture.

Thank you very much and have a good day.