

Advanced Design of Steel Structures
Dr. Srinivasan Chandrasekaran
Department of Ocean Engineering
Indian Institute of Technology, Madras

Lecture - 55
LTB example problem

(Refer Slide Time: 00:20)

Lecture 55

- LTB - example

imposed load (w) = 6.25 kN/m
dead load (9.58 kN/m)

Span = 5.7m

Beam section: UKB 356x171x51
steel class: S 275 grade

NPTEL

Friends, welcome to the lecture 55, where we are going to do an example problem for Lateral Torsional Buckling. We will take a simply supported beam for a span of 5.7 meters subjected to two kinds of loads. One load is a dead load, which is about 9.58 kilo newton per meter.

And over and above we have an imposed load which is uniform distributed, and this is imposed load otherwise called as live load of intensity 6.25 kilo newton per meter. We will call this end as A, this end as B. Now, we are going to use a specific kind of trial beam section. The beam section what we chose is UK beam 356 by 171 by 51, which is S 275 grade.

(Refer Slide Time: 02:23)

(1) To estimate the max BM (M_{ED})
 a) factored loads $w = (1.35 \times 9.58) + (1.5 \times 6.25)$
 $= 22.31 \text{ kN/m}$
 $M_{ED} = \frac{wL^2}{8} = \frac{22.31 \times 5.7^2}{8} = 90.6 \text{ kNm}$
 $r = 355 \text{ mm}$
 $b = 171.5 \text{ mm}$
 $t_w = 7.4 \text{ mm}$
 $t_f = 11.5 \text{ mm}$
 $r = 10.2 \text{ mm}$
 steel grade, S275, $\sigma_y = 275 \text{ N/mm}^2 = f_y$
 max thickness, $11.5 \text{ mm} < 40 \text{ mm}$ ok (E = 210 GPa)

So, for the benefit of the users, let us show the cross section of the beam. This one the cross section of the UK beam 356, 171 and 51. Based on this, let us write down certain dimensions. Now, the first step is to estimate the maximum bending moment. We call this as M_{ED} . Let us see what the factored loads w are, which is going to be 1.35 into 9.58; this value plus 1.5 into 6.25.

So, I am using a different factor for live load and dead load which is going to be 22.31 kilo newton per meter. So, I can quickly find out M_{ED} as $w l^2$ by 8 because simply supported section. So, it is going to be 22.31 into 5.7 square by 8, which will give me 90.6 kilo newton meter, this is my demand.

Now, for the section shown on the screen the properties are like this; h is 355, that is why the section is called 356 355; b which you see here is 171.5. These are all millimeters friends; t_w which you see here thickness of the web is 7.4 millimeter, t_f thickness of the flange which you see here is 11.5 millimeters, r is 10.2 millimetre which is actually the radius of curvature being used for fabricating the ends of this rolled steel section.

These are available in the standard tables. They are referred as UK steel sections. Similar also available Indian code. We have Indian code sp 61 which will give you the section properties and so on. So, the steel grade which is being used is S 275, which means σ_y is 275. Yield strength is 275 newton per mm square. We will use f_y because that is a symbol used in Euro code.

Let us check quickly, what is the maximum thickness which is permitted for this kind of steel. The maximum thickness in the section available is 11.5. This is lesser than 40 that is a check, that is a check. So, this is according to I will write here this according to EN 1993-1-1 2005; table 3.1, that is what I have used.

(Refer Slide Time: 06:20)

section classification

$$\epsilon = \sqrt{235/f_y} = 0.92 \quad (\text{Table 5.2})$$

Outstanding flange (uniform compres)

$$c = \frac{(b - t_w) - 2r}{2} = \frac{(171.5 - 7.4) - (2 \times 10.2)}{2} = 71.85 \text{ mm}$$

$$c/t_f = \frac{71.85}{11.5} = 6.25 < 9\epsilon \quad (\text{Class 1 Element (Table 5.2)})$$

Internal compression part (webs in bending)

$$c = (h - 2t_f - 2r) = 355 - (2 \times 11.5) - (2 \times 10.2) = 311.6 \text{ mm}$$

$$c/t_w = 311.6 / 7.4 = 42.1 < 72\epsilon \quad (\text{Class 1 Element})$$

Now, let us come to section classification. Section classification depends on square root of 235 by f_y which comes to 0.92. This is as per table 5.2, I am using the same code. Let us talk about the outstanding flange. The outstanding flange which is under uniform compression.

We will work out the c value is b minus t_w minus $2r$ by 2 , which will be b is 171.5, thickness of the web is 7.4 minus 2 times of 10.2, that is the radius given; divided by 2 which gives me 71.85 millimetres. So, a factor c by t_f is 71.85 by 11.5 which is 6.25 which is less than 9 epsilon. So, this gives me that it is class 1 element. This is as per table 5.2 of the code. So, it is a class 1 element now.

Let us talk about the internal compression member. So, now, we are considering web in bending. Bending can also have compression, bending compression. Let us compute that for c is equal to h minus $2t_f$ sorry $2t_f$ minus $2r$ which is going to be 355 minus 2 times of 11.5 minus 2 times of 10.2 which gives me 311.6 millimetres. Now, c by t_w which is 311.6 divided by 7.4 is 42.1 which is less than 72 epsilon, which also guarantees and classifies as class 1 element. This is as per table 5.2.


(Refer Slide Time: 09:28)

Now, the section used for trial is Class 1 section

(b) To compute $M_b R_d$, $M_{b, Rd}$
Clause 6.2.4, EN 1993-1-1 (2005).

$$M_{b, Rd} = \frac{W_{pl,y} f_y}{\gamma_{M0}} = \frac{(896 \times 10^3) 275}{(1.0)} \times 10^{-6} = 246.4 \text{ kNm}$$

Since Capacity ($M_b R_d$) $>$ demand (M_{Ed})
 $(246.4) \text{ kNm} > 90.6 \text{ kNm}$
the section is safe against bending



Having said this, we are now using; so, now, the section used for trial is class 1 section, we have we have clarified that. Now, let us do the second step which is to compute the moment of resistance which is $M_b R_d$. We will use class 6.2.4 of Euro code 1993-1-1 2005, according to this $M_b R_d$ is given by $W_{pl,y}$ into f_y by γ_{M0} . So, we will say 896 is capital W, 275 is my yield strength. And γ_{M0} is taken as 1.0 to 10 power minus 6.

So, so many kilo newton meter which gives me 246.4 kilo newton meter. Now, friends please note, the capacity which is moment of resistance is higher than the demand, which is M_{Ed} , what we have used the symbol M_{Ed} . So, one can say. So, since that is 260 sorry 246.4 is greater than 90.6 kilo newton meter, we can say the section is safe against bending. We have checked it; we are now checking a trail section.

(Refer Slide Time: 11:57)

(c) check against LTB

To compute the elastic critical moment (M_{cr})

- support condition: simply supported - effective length factor $k_w = k_z = 1$


load applied: top flange, Z_g will reduce the section by half 0.5

$$M_{cr} = C_1 \frac{\pi^2 EI_z}{(k_z L)^2} \left\{ \left[\left(\frac{k_z}{k_w} \right)^2 \frac{I_w}{I_z} + \frac{(k_z L)^2 GI_T}{\pi^2 EI_z} + (C_2 Z_g - C_3 Z_j)^2 \right]^{0.5} - (C_2 Z_g - C_3 Z_j) \right\}$$

$Z_g = h/2 = \frac{355}{2} = 177.5 \text{ mm}$

C_1 (Table) 1.12

C_2 (Table) 0.45



Now, let us check against LTB. So, to check against LTB, we are going to check the elastic critical moment. Now, to compute the elastic; so, based on the support conditions and the loading applied at the top flange level. So, now, support conditions we should say simply supported, load application top flange. We can calculate M_{cr} . We already derived this equation.

M_{cr} is given by that is three parameter formula

$$C_1 \frac{\pi^2 EI_z}{(k_z L)^2} \left\{ \left[\left(\frac{k_z}{k_w} \right)^2 \frac{I_w}{I_z} + \frac{(k_z L)^2 GI_T}{\pi^2 EI_z} + (C_2 Z_g - C_3 Z_j)^2 \right]^{0.5} - (C_2 Z_g - C_3 Z_j) \right\}. \text{ This equation}$$

only we wrote the previous lecture, please refer back. So, now we have to use the figures and tables.

So, the conditions are the beam is simply supported. Therefore, the effective length factor which is k_w which is also equal to k_z which is 1.0. Now, since the load is separated in top flange Z_g will reduce the section by half. So, what does it mean? Z_g will be actually h by 2, which is 355 by 2, which is 177.5 millimeters. C_1 from table which you already said is 1.12, C_2 from table is 0.45.

(Refer Slide Time: 15:47)

$I_z = 968 \text{ cm}^4$
 $I_w = 0.286 \text{ dm}^4$ (1 dm = 10 cm)
 $I_t = 23.8 \text{ cm}^4$
 $E = 2.1 \times 10^5 \text{ N/mm}^2$
 $G = 77 \text{ kN/mm}^2$

$M_{cr} = 121.9 \text{ kNm}$

Then, from the table we will also know I_z for the section which is 968 centimeter to power 4, I_w is 0.286 d m to the power 6, this is given in UK steel tables. I am just taking the value directly from there. So, 1 decimeter is 10 centimetre. Let us have this connection here. I_t is 23.8 centimeter 4 and E of course, is 2.1×10^5 newton power square. And G is 77 kilo newton per m square.

Now, we know all the values. See here we wanted to know I am just checking we wanted to know E sorry C_1 , we have E we have, I_z we have k_z , we have L of course, we have. I_w we have, I_t we have, G we have, I_t we have, Z_{gc2} we have, Z_{gc} we have. So, you can compute these values directly and I will get M_{cr} as 121.9 kilo newton meter,.

(Refer Slide Time: 18:06)

To estimate the buckling factor, χ_{LT}

slenderness parameter, $\bar{\lambda}_{LT} = \sqrt{\frac{W_y f_y}{M_{cr}}}$

$$= \sqrt{\frac{(896 \times 10^3)^{275}}{121.9 \times 10^6}} = 1.423$$

Consider the section as hot-rolled, use the special case method

$\left(\frac{h}{b}\right) = \frac{355}{171.5} = 2.05 > 2$, use buckling curve C

for curve C, $\alpha_{LT} = 0.49$

So, the next step is to estimate the buckling factor, which is ψ_{LT} . This depends on slenderness. So, slenderness parameter which is $\bar{\lambda}_{LT}$ is given by square root of $W_y f_y$ by M_{cr} . We already have this value see here W_y we have, f_y we have, now M_{cr} also we have. So, let us substitute that. It is going to be square root of 896 into 10 power 3 into 275 divided by 121.9 10 power 6. I get this value as 1.423, you can please check this.

So, now we consider this section as hot rolled. So, we will use this special case method. There are two methods by which we can do. So, we know from the table for h by b which is equal to 355 by 171.5 which is 2.05 greater than 2, we should use buckling curve C, am I right?

We have seen this in the last lecture. For rolled section for h by b more than 2, what buckling curve we should use? A special case method. Please refer to your notes. Now, from the buckling curve for curve C, we can directly find α_{LT} as 0.49.

(Refer Slide Time: 20:31)

$$\phi_{LT} = 0.5 \left[1 + \alpha_{LT} (\bar{\lambda}_{LT} - \bar{\lambda}_{LT,0}) + \beta \bar{\lambda}_{LT}^2 \right]$$

$$\phi_{LT} = 0.5 \left[1 + 0.49(1.423 - 0.4) + 0.75 \times 1.423^2 \right] = 1.510$$

$$X_{LT} = \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \beta \bar{\lambda}_{LT}^2}} \leq \min\left(1.0, \frac{1}{\bar{\lambda}_{LT}^2}\right)$$

$$X_{LT} = 0.42 = \frac{1}{1.51 + \sqrt{1.51^2 - 0.75 \times 1.423^2}} = \frac{1}{1.51 + 1.423} = 0.42$$


Then we can compute ϕ_{LT} which is given by 0.5 times of 1 plus α_{LT} times of $\bar{\lambda}_{LT}$ minus $\bar{\lambda}_{LT,0}$ plus beta $\bar{\lambda}_{LT}$ square of LT . So, let us work out ϕ_{LT} in this case 0.5 times of 1 plus 0.49 1.423 minus 0.4 plus 0.75 1.423 square, which gives me 1.510. So, now, I can say ψ_{LT} is given by 1 by ϕ_{LT} plus square ϕ_{LT} minus beta $\bar{\lambda}_{LT}$ square LT . And which should be the minimum of 1.0 and 1 by $\bar{\lambda}_{LT}$ square LT .

So, now I get ψ_{LT} as 1 by 1.51 plus square of 1.51 square minus 0.75, that is the beta value into 1.423 square, which comes to 0.42 which is less than 1 and also 1 by $\bar{\lambda}_{LT}$ square LT is 1 by 1.423 square which is 0.493, right. So, the value what we got is 0.42. So, I can now say ψ_{LT} can be taken as 0.4. It satisfies both the conditions no.

(Refer Slide Time: 23:01)

The beam is under Udl; for this BM distn,
from Table $k_c = 0.94$

f-factor

$$f = 1 - 0.5(1 - k_c) \left[1 - 2(\bar{\lambda}_{cr} - 0.8)^2 \right] \leq 1$$
$$= 1 - 0.5(1 - 0.94) \left[1 - 2(1.423 - 0.8)^2 \right]$$
$$= 0.993 < 1$$


Now, we have the beam under uniform distributed load. So, for this load distribution or specifically, for this bending moment distribution, from the table which we discussed in the last lecture, I can find k_c as 0.94. Please see the table. Now, I want to calculate the f factor because I am using the second method.

So, f factors required; f is given by 1 minus 0.5 times of 1 minus k_c of 1 minus twice of $\bar{\lambda}_{cr}$ minus 0.8 square should be less than or equal to 1. Let us check 1 minus 0.5 times of 1 minus 0.94 of 1 minus twice of 1.423 minus 0.8 the whole square, we get this value as 0.993 which is less than 1. So, f factor is now computed.

(Refer Slide Time: 24:28)

$$M_{oR} = M_{bRd}$$
$$M_{bRd} = \frac{\chi_{LT}}{f} W_{pl,y} \frac{f_y}{\gamma_{M1}}$$
$$= \left(\frac{0.42}{0.993} \right) (896 \times 10^3) \left(\frac{275}{1} \right) = \underline{104.2 \text{ kNm}}$$

The demand, $M_{ED} = 90.6 \text{ kNm}$
The capacity, $M_{bRd} = 104.2 \text{ kNm}$

$\therefore M_{ED} < M_{bRd}$
Safe against LTB
(no additional lateral restraints are required)

Therefore, we can then find the M_{oR} which is actually M_{bRd} . M_{bRd} is given by χ_{LT} by $f W_{pl,y}$ by γ_{M1} . We will substitute 0.42 that is χ_{LT} , is it not? 0.42 f factor is 0.993 and W_p is 896×10^3 into 275 by 1 . We already said γ_m is 1 for this kind of distribution. So, this value comes to be 104.2 kilo newton meter.

So, now the demand is M_{ED} which is 90.6 kilo newton meter. The capacity M_{bRd} which is 104.2 kilo newton meter. Since the demand is lesser than the capacity, the beam is safe against lateral torsional buckling. So, we are just checking whether the beam is safe. So, what is the comment here? No additional lateral restraints are required. Otherwise, you have got to provide lateral restraints as we discussed in the previous lectures.

(Refer Slide Time: 26:39)

Using IS code (IS 800-2007)

- section ISMB 400 @ 61.6 kg/m.
- The beam is simply supported over a span of 8m
- total moment under UDL = 50 kNm
- fy 250 grade is to be used

Check the beam for LTB

Let us solve the same problem using IS code. We are using IS 800-2007, that is the steel design code by checking the section. So, we will take the section as ISMB 400 at 61.6 kg per meter. So, let us say the beam is simply supported over a span of 8 meters. The total moment under uniform distributed load is 50 kilo newton meter, fy 250 grade is to be used. So, check the beam for lateral torsional buckling that is the question asked. Let us do that.

(Refer Slide Time: 28:08)

Section properties

$$A = 78.46 \text{ cm}^2 = 7846 \text{ mm}^2$$

depth of section, $h = 400 \text{ mm}$

$$t_f = 16 \text{ mm}$$
$$b_f = 140 \text{ mm}$$
$$t_w = 8.9 \text{ mm}$$
$$I_{yy} \text{ (Minor axis)} = 622.1 \text{ cm}^4 = 622.1 \times 10^4 \text{ mm}^4$$
$$I_{xx} \text{ (Major axis)} = 20458.4 \text{ cm}^4 = 20458.4 \times 10^4 \text{ mm}^4$$

Let us first see the section properties. Area of the cross section which is 78.46 centimeter square which is 7846 millimeter square, depth of the section h 400 mm, thickness of the

flange 16 mm, breadth of the flange 140 mm, thickness of the web 8.9 mm. This can be seen from steel tables. Moment of inertia about the minor axis also given in the table which is 622.1 centimeter 4 which is 622.1 10 power 4 mm. I xx which is about the major axis which is 20458.4 which is 20458.4 into 10 power 4 mm 4.

(Refer Slide Time: 29:37)

Section Classification

$f_y = 250 \text{ N/mm}^2$

Table 2 of IS 800-2007, $\epsilon = \sqrt{\frac{250}{f_y}} = 1$

$\left(b = \frac{b_f}{2}\right)$ $\leftarrow \frac{b}{t_f} = \frac{\frac{140}{2}}{16} = 4.375 < 9.4 \epsilon$ (for $\epsilon = 1$)

$\frac{d}{t_w} = \frac{400 - (2 \times 16)}{8.9} = 41.35 < 85 \epsilon$

$d = d_w$

Classified as PLASTIC SECTION

$\frac{h}{b_f} = \frac{400}{140} = 2.86 > 1.2$

Then we will do the section classification. The first step is section properties, we know f_y is 250 newton per meter square. Let us look into table 2 of IS 800 2007. So, epsilon is given by square root of 250 by f_y which is equal to 1 and b by t_f in our case is going to be over an. So, 140 by 2 divided by 16 which is 4.375 less than 9.4 epsilon 4 epsilon equal to 1,. And d by t_w which is 400 minus 2 into 16 divided by 8.9, which is 41.35 which is less than 85 epsilon,.

So, friends in this equation if you note this b actually is b_f by 2. Look at figure 2 of IS 800. And of course, d is d_w , only the web. So, now, based on these data we can classify the section as , classify as plastic section. Furthermore, h by b_f is 400 by 140 which is 2.86 greater than 1.2 and t_f is 16 mm less than 40 mm.

(Refer Slide Time: 32:12)

$G_t = 16000 \text{ N/mm} < 40000$ (Table 10, NPTEL)

buckling class about XX axis is 'a'
 " " about YY axis is 'b'

M_{cr}
 Symmetric
 Prismatic
 Simply supported

$$M_{cr} = \sqrt{\frac{\pi^2 E I_y}{L_{eff}^2} \left[G I_t + \frac{\pi^2 E I_w}{L_{eff}^2} \right]}$$
 - class 8.2.2

$E = 2.1 \times 10^5 \text{ N/mm}^2$
 $\mu = 0.3$

$G = \frac{E}{2(1+\mu)} = \frac{2.1 \times 10^5}{2(1+0.3)} = 80.77 \times 10^3 \text{ N/mm}^2$

So, from Table 10 of the code buckling class about XX axis is a and buckling class about YY axis is b. So, then the next step is to find out M_{cr} that is elastic lateral buckling moment. Now, the section is symmetric. It is prismatic and it is simply supported, because we should have the bending moment distribution also these are the conditions.

For these conditions, M_{cr} is given by square root of $\pi^2 E I_y$ by L_{eff}^2 square of $G I_t$ plus $\pi^2 E I_w$ by L_{eff}^2 square. This is as per class 8.2.2 2.21, sorry. So, in this let us substitute the values. E is 2.1×10^5 . So, we have E μ is 0.3, therefore, G is E by twice of $1 + \mu$ which gives me 2.1×10^5 by twice of $1 + 0.3$. I get this value as 80.77×10^3 . So, I have G , I_y already we have. Let us see what is I_y 622.1.

(Refer Slide Time: 34:55)

$I_y = 622.1 \times 10^4 \text{ mm}^4$
 Effective length, $L_e = 1.0L = 9\text{m}$ (Table 15, class 8.3 for torsional-fully restrained but warping not restrained in both the flanges)
 $I_{\tau} = \text{Torsional constant} = \sum \frac{b_i^3 t_i^3}{3}$ for open section
 $I_{\tau} = 2 \times \frac{(140 \times 16^3)}{3} + \frac{(368 \times 8.9^3)}{3} = 468.77 \times 10^3 \text{ mm}^4$
 Warping constant, $I_w = (1 - \beta_f) \beta_f I_y^2 / r_y^2$ for I section
 $\beta_f = \frac{I_{afc}}{I_{afc} + I_{aft}}$ (I_{afc}, I_{aft} — MOI of tension & comp flange about the minor axis)
 $I_{afc} = \frac{16 \times 140^3}{12} = 3.66 \times 10^6 \text{ mm}^4 = I_{aft}$

So, I_y we have, 622.1×10^4 . Effective length which is L_e , is 1 point L which is 8 meters. So, this is according to table 15 class 8.3 for torsional fully restrained, but warping not restrained in both the flanges. For this condition you have taken this effectively. So, I_{τ} torsional constant which is summation of $b_i^3 t_i^3$ by 3, this is for open sections.

So, let us find out for our problem, I_{τ} twice of 140 into 16 cube by 3 plus 368 into 8.9 cube by 3, which comes to be 468.77×10^3 mm⁴, warping constant. So, let say I_{τ} we have, L_e we have, we want to find I_w . So, warping constant I_w is $(1 - \beta_f) \beta_f I_y^2 / r_y^2$, this is for I sections.

So, now what is β_f ? β_f is I_{afc} by $I_{afc} + I_{aft}$ where I_{afc} and I_{aft} are moment of inertia of tension and compression flanges about the minor axis of the section. Let us compute I_{afc} that is very easy. So, I_{afc} about minor axis. So, $p t^3$ by 12, I can now say this is going to be 3.66×10^6 mm⁴. So, we have the same as I_{aft} tension and compression flange is minor axis moment of inertia is same.

(Refer Slide Time: 38:30)

h_y - distance b/w the shear center & flange of the x-z

$$h_y = h - \left(\frac{t_f}{2}\right) = 400 - 16 = 384 \text{ mm}$$

$$\beta_f = \frac{I_{fc}}{I_{fc} + I_{ft}} = \frac{3.66 \times 10^4}{(3.66 \times 10^4)^2} = 0.5$$

$$I_w = (1 - \beta_f) \beta_f I_y h_y^2$$

$$= (1 - 0.5)(0.5) \times (622.1 \times 10^4) (384)^2$$

$$= 2.293 \times 10^{11} \text{ mm}^4$$

Having said this, let us now find what is h_y which is actually the distance between the shear center of two flanges of the cross section. So, h_y in my problem will be h minus t_f by 2 into 2 which will be 400 minus 16 which is 384, β_f which is I_{fc} by I_{fc} plus I_{ft} which 3.66×10^4 power 6 by 3.66×10^4 power 6 into 2 which is 0.5.

Now, let us work out I_w which is given by 1 minus β_f β_f I_y h_y square. Let us substitute them. 1 minus 0.5 , 0.5 I_y 621.1 that is here, 622.1×10^4 , h_y is here 384 square which gives me the value as $2.293 \times 10^{11} \text{ mm}^4$. Now, I can find M_{cr} .

(Refer Slide Time: 40:40)

$$M_{cr} = \sqrt{\frac{\pi^2 E I_f}{L_{cr}^2} \left[G I_t + \frac{\pi^2 E I_w}{L_{cr}^2} \right]} - \text{clau 1 p. 2-4}$$

$$= \sqrt{\frac{\pi^2 \times (2.1 \times 10^5) \times (622.1 \times 10^4)}{(8000)^2} \left[(80.77 \times 10^3) \times (468.77 \times 10^3) + \frac{\pi^2 \times (2.1 \times 10^5) \times (2.293 \times 10^{11})}{(8000)^2} \right]}$$

$$M_{cr} = 95.51 \times 10^6 \text{ Nmm} = 95.51 \text{ kNm}$$

M cr let us copy this equation of M cr back again is given by this equation. Let us substitute which will be equal to square root of pi square 2.1 10 power 5 that is my E value, I is 622.1 10 power 4. This is 8000 square. I am doing it in millimetres, in newtons. Then 80.77 10 power that is my G value see here , that is my G value I tau is 468.77 10 power 3 that is my tau value plus pi square 2.1 10 power 5.

I warping constant we already have it here 2.293 , divided by 8000 square. So, I get M cr as 95.51 into 10 power 6 newton mm which will become 95.51 kilo newton meter,.

(Refer Slide Time: 43:04)

design BM

$$M_d = \beta_b Z_p f_{bd} \quad (\text{class 8.2.2 of IS 800-2007})$$

$\beta_b = 1$ plastic sections
 Z_p - plastic section modulus
 f_{bd} - design comp bending stress = $\frac{\chi_{LT} f_y}{\gamma_{m0}}$

$$\chi_{LT} = \frac{1}{\left[\phi_{LT} + \sqrt{\phi_{LT}^2 - \lambda_{LT}^2} \right]}$$

Now, let us compute the design bending moment. The design bending moment is M d which is beta b Z p f bd. This is as per class 8.2.2 of IS 800. So, what is beta b? Which is 1 for plastic sections. We already declared this the plastic section. What is Z p? Is the plastic section modulus, and f bd? Is a design compressive bending stress which is actually equal to psi LT f y by gamma m naught. So, let us say what is psi LT, we already know this phi LT plus phi LT square minus lambda square LT,.

(Refer Slide Time: 44:29)

$$\phi_{LT} = 0.5 \left[1 + \alpha_{LT} (\lambda_{LT} - 0.2) + \lambda_{LT}^2 \right]$$

$$\alpha_{LT} \text{ imperfection factor} = 0.21 \text{ (rolled section)}$$

$$\lambda_{LT} = \sqrt{\frac{\beta_b Z_p f_y}{M_{cr}}} \leq \sqrt{\frac{1.2 Z_e f_y}{M_{cr}}}$$

$$Z_p = 1176163.26 \text{ mm}^3 \text{ (ISMB 400)}$$

$$Z_e = 1022.9 \times 10^3 \text{ mm}^3$$

$$\lambda_{LT} = \sqrt{\frac{1 \times (1176163.26) (250)}{95.51 \times 10^6}} \leq \sqrt{\frac{1.2 \times (1022.9 \times 10^3) (250)}{95.51 \times 10^6}}$$

$$= 1.75 \leq 1.79 \text{ (OK)}$$

SP 6 (U) Steel Table

Φ_{LT} is 0.5 times of 1 plus α_{LT} into λ_{LT} minus 0.2 plus α_{LT} bar λ_{LT} square α_{LT} is the imperfection factor which is 0.21 for rolled sections given in the code. Let us calculate λ_{LT} which is actually equal to square root of $\beta_b Z_p f_y$ by M_{cr} which is less than or equal to square root of $1.2 Z_e f_y$ by M_{cr} . So, let us say what is Z_p ; 1176163.26. This is for ISMB 400 Z_e is 1022.9×10^3 mm cube,.

We can refer these values from SP 6 1 steel tables. So, now, λ_{LT} will be 1 into 1176163.26 into 250 by 95.51×10^6 which will become which should be less than or equal to 1.2 times of 1022.9×10^3 of 250 by 95.51×10^6 . So, now we get this value as 1.79 less than or equal to one point this is 75 79. So, it is ,. Once I have this value we can substitute back in ϕ_{LT} and cut ϕ_{LT} , let us do that.

(Refer Slide Time: 47:20)

$$\phi_{LT} = 0.5 \left[1 + \alpha_{LT} (\chi_{LT} - 0.2) + \lambda^2_{LT} \right]$$

$$= 0.5 \left[1 + 0.21 (1.75 - 0.2) + 1.75^2 \right] = 2.194$$

$$\chi_{LT} = \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \lambda^2_{LT}}} = \frac{1}{2.194 + \sqrt{2.194^2 - 1.75^2}}$$

$$= 0.284$$

$$f_{bd} = \frac{\chi_{LT} f_y}{\gamma_{m0}} \quad (\gamma_{m0} = 1.1 - \text{partial safety factor of member to resist buckling})$$

$$= \frac{0.284 \times 250}{1.1} = 64.616 \text{ N/mm}^2$$

So, therefore, ϕ_{LT} is going to be 0.5 times of 1 plus α_{LT} times $(\chi_{LT} - 0.2)$ plus λ^2_{LT} which will be 0.5 times of 1 plus 0.21 that is what my α_{LT} is 0.21 into 1.75 minus 0.2 plus 1.75 square, which comes to 2.194. So, now can calculate this value as 1 by ϕ_{LT} plus ϕ_{LT} square minus λ^2_{LT} which will be 1 by 2.194 plus 2.194 square minus 1.75 square which comes to 0.284. Now f_{bd} bending stress in compression is f_y by γ_{m0} .

Now, γ_{m0} is equal to 1.1 which is the partial safety factor of member to resist buckling. So, let us substitute that which will become 0.284 into 250 by 1.1 which comes 64.616.

(Refer Slide Time: 49:53)


The design bending strength is governed by the lateral torsional buckling

$$M_d = \beta_b Z_p f_{bd}$$
$$= (1.0) (1176163.26) \times 64.616 = 75.99 \approx 76 \text{ kNm}$$

The applied moment is 50 kNm

Capacity is 76 kNm

- Safe against LTB
- No additional restraints are required



So, now the design bending strength is governed by the lateral torsional buckling. So, M_d is $\beta_b Z_p f_{bd}$ which is $1.0 \times 1176163.26 \times 64.616$ which is 75.99 or 76 kilo newton meter. The applied moment is 50 kilo newton meter, the capacity is 76 kilo meter. Therefore, we can say the beam is safe against lateral torsional buckling no additional restraints are required.


(Refer Slide Time: 51:31)

Summary

- 2 Examples of check the design for LTB
- (1) Euro code
- (2) Indian code

(Capacity)_{des} > Demand

Safe against LTB



So, friends, in this lecture, we learnt two examples of checking the design for lateral torsion work. 1, we did the Euro code, 2nd one we did with Indian code. In both the cases, since the

capacity of the section is larger than the demand of the section. So, we declare they are safe against LTB.

So, the procedure is very straight forward. It is well illustrated in both the course, equations and supporting tables are available to you. So, only I got to practice more number of problems. So, take some sample sections, work it out and see whether the chosen section is safe against lateral torsional buckling,.

Thank you very much and have a good day.